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ANALYSIS OF CAR WASTE HEAT RECOVERY SYSTEM UTILIZING THERMOELECTRIC GENERATOR

This paper presents a calculation algorithm for a thermoelectric generator fitted in the exhaust system of a combustion engine. The viability of the presented calculation method was verified on an actual combustion engine. The calculations were performed for a BMW engine, and the generator design was based on a prototype from the same manufacturer. The paper includes calculations of the thermal cycle and of the parameters of exhaust gases from the engine. Subsequent calculations cover heat transfer from exhaust gases to the thermoelectric module and the amount of electric energy obtained from a series of modules. In the last part, the focus is on the influence of engine speed on the performance of the thermoelectric generator.

INTRODUCTION

Modern cars are equipped with a number of features which enhance the comfort of both the driver and the passengers. These features include air conditioning, satellite navigation, power windows, seat heating etc., and all of them are powered by an alternator. Lighting system is part of standard equipment in every car. Headlamps are fitted with two low beam bulbs and two high beam bulbs. The low beam lamps may draw between 110 W and 120 W of electric energy. The high beam lamps may draw additional 120 W, and fog lamps require another 110 W of power [1]. If the list is to include position lamps, turn signals and a radio set, the total amount of energy demand may exceed 400 W. Although a growing number of comfort features entails increased energy consumption, a great part of this equipment has been optimized, for example by substituting traditional bulbs with LED lamps. Fig. 1 shows electricity demand in cars, based on [2, 3, 4].



Fig. 1. Electricity demand in vehicles by size [based on 2, 3, 4]

Alternator is a device which receives mechanical power from the crankshaft. Due to its limited efficiency, the alternator consumes a significant portion of energy produced by the engine. Lowering energy consumption in vehicles has thus become the focus of much research. However, reductions in this field are possible only to some extent, as a certain level of comfort and safety must be preserved, for instance with regard to the lighting system of a vehicle. Therefore, researchers and design engineers are seeking for alternative solutions which allow limiting the use of energy taken from the crankshaft

to power on-board electrical devices. The most promising solutions [5, 6] include Organic Rankine Cycle systems and thermoelectric generators, which are the focus of this paper.

1. THERMOELECTRIC GENERATORS

Thermoelectric generators may utilize waste energy from exhaust gases to limit the power output of the alternator. Their principle of operation is based on Seebeck effect, which is widely described in literature [7]. The generator is designed so as to capture thermal energy from exhaust gases and transform it into useful electrical energy.

Various generator designs represent a number of different structures and shapes. Typically, the system comprises a metal duct surrounded by thermoelectric modules and a radiator whose function is to maintain a comparatively low temperature on the cold side of the thermoelement. The cooling agent is typically a cooling liquid from the combustion engine system. Some solutions incorporate additional bypass lines which allow exhaust gases to be diverted around the TEG in the case when pressure values are excessively high in the exhaust system. The electric energy provided by the generator is managed by a controller and by a system which processes unstable voltage into nominal voltage.

As the cost of conducting research into thermoelectric generators is high, first models of such devices were used in the space, military and aviation industries.



Fig. 2. A view of a hexagonally arranged electric generator [8]

One example of a hexagonal arrangement of thermoelements is provided in a generator described in [8], fitted in the exhaust system of a 1.3 dm³ JTD engine. The generator has variable geometry fins. This solution is supposed to positively influence temperature distribution along the heat exchanger.

The TEG body was made of aluminum, as this material has very high thermal conductivity and is corrosion resistant. The modules were positioned between the body and individual radiators, and were fixed with press bolts (Fig. 2). The parameters of the generator are shown in Table 1.

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Parameter	Value
Overall dimensions	280 mm x 555 mm
Heat exchanger dimensions	110 mm x 311 mm
Internal heat exchange surface	0.574 m ²
Heat exchanger material	aluminum alloy
Number of thermoelectric modules	24
Nominal power of a single TEM	7 W (<i>T_C</i> = 50°C, <i>T_H</i> = 175°C)
Total nominal power of the thermoelectric modules	168 W
Open circuit voltage (for T _{in} = 250°C)	280 V
Maximum TEM operating temperature	200°C
Total mass	~ 1260 g

A concept of thermoelectric generator designed at the Institute of Vehicle Concepts, German Aerospace Center (DLR) in Stuttgart was presented by the BMW Group in [2]. The model was equipped with 24 thermoelectric Bi₂Te₃ modules and with 4 cooling liquid heat exchangers, arranged in layers. a diffuser inside the exhaust gas pipe was installed in order to regulate the flow of exhaust gases. The system was implemented in a BMW 535i car with a 3.5-liter spark ignition engine. The modules are capable of withstanding maximum temperature of 250°C and therefore the system included an additional bypass line fitted with flap valves to ensure protection against over-heating. The cold side of the thermoelement was cooled with a coolant circulating in a separate loop. The solution is shown in Fig. 3.



Fig. 3. Schematic diagram of the BMW-designed TEG [2]

The system was able to collect 200 W of electric energy at vehilce constant speed of 130 km/h. The researchers claim that in the future the power output may be significantly increased by using improved thermoelectric elements. At 250°C, the modules used in the actual generator showed very low value of quality factor ZT= 0.4. According to the researchers, if this value can be increased to 0.85, the power output could reach 600 W [2]. More information on the current applications of thermoelectric generators can be found in [9].

Knowledge on the operating parameters of the engine and its components is vital during the design stage. This knowledge may contribute to the designing of a highly efficient system which recovers energy from waste exhaust gases and transforms it into useful energy. Although researchers and designers in large car manufacturing companies have been working on thermoelectric generators for several years, no calculation algorithm for such systems has been presented in literature so far. Moreover, many publications on this issue do not include all of the parameters which describe the process and the system of waste energy recovery. This paper aims at addressing the above issue by providing a detailed case study.

2. THE MATHEMATICAL MODEL

The mathematical model uses the known thermodynamic parameters of the engine's working medium, geometrical dimensions of the system's components and optimization relationships of thermoelectric modules.

The engine used in the calculations was a four stroke, six cylinder, in-line, gasoline powered BMW M50B20 model from a medium size car, whose parameters are shown in Table 2. Due to the research results obtained by the BMW Group, a decision was made to base the model on a design similar to the design described in [2].

BMW M50 B20							
Parameter	Symbol, unit	Value					
Configuration	i	6					
Swept volume	V _{sk} , dm ³	2					
Power	Ps, kW	110 at 6000 rev/min					
Torque	<i>M₀</i> , Nm	190 at 4700 rev/min					
Compression ratio	3	10					
Number of strokes per one work cycle	Т	2					
Fuel	-	Gasoline					

Tab. 2. Parameters of the BMW M50B20 engine

2.1. Stoichiometric calculations

In order to determine the flow of exhaust gases generated by a combustion engine, air excess factor λ should be assumed at 1.05 and fuel calorific value should be calculated. Based on literature [10], the ratio between hydrogen and carbon was assumed at w_h =0.15 and w_c =0.85, respectively. When the calorific values of those elements are known, the resultant calorific value of gasoline is calculated from (1).

$$W_d = 33910 \frac{kJ}{kg} \cdot w_c + 121400 \frac{kj}{kg} \cdot w_h, \frac{kj}{kg}$$
(1)

Table 3 shows intake fuel and air parameters used in the calculations. Under assumption that combustion is complete and perfect, combustion stoichiometry may serve to determine actual air consumption. For this purpose, the amount of oxygen per each kilogram of fuel was calculated from (2).

$$N_{O_2} = \left(\frac{w_c}{m_c} + \frac{w_h}{w_h}\right), \frac{kmol}{kg}$$
(2)

The amount of air per each kilogram of fuel may be calculated from (3), and the actual amount of air per each kilogram of fuel is described by equation (4).

$$N_{air} = \frac{N_{O_2}}{0.21} , \frac{kmol}{kg}$$
(3)

$$N_{air.act} = \lambda \cdot N_{air}, \frac{kmol}{ka}$$
(4)

The calculation of actual air volume (5) was based on Avogadro's law.

$$V_{air.act} = 22.4 \frac{m^3}{kmol} \cdot N_{air.act} , \frac{m^3}{kg}$$
(5)

Tab. 🤇	3.	Parameters	of fuel	and	air at	the	inlet	to	the	engine
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W_d	N _{O2}	N _{air}	N _{air.act}	V _{air.act}
kJ/kg	kmol/kg	kmol/kg	kmol/kg	m³/kg
47034	0.108	0.516	0.542	12.133

Further calculations were performed for several rotational speeds of the combustion engine (1500, 1750, 2000, 2250, 2500, 2750, 3000, 3250, 3500). Change in the rotational speed effects a change in the amount of exhaust gases produced, which in turn affects the flux of wasted energy. The range of rotational speeds was selected so as to reflect typical operating characteristics of a combustion engine.

Relationship (6) describes the amount of air at the inlet to the combustion chamber, while equation (7) describes the mass flow rate of the combusted fuel.

$$\dot{V}_{air} = \frac{n \cdot V_{sk}}{\tau}, \frac{m^3}{s}$$

$$\dot{m}_{fuel} = \frac{\dot{V}_{air}}{\tau}, \frac{kg}{s}$$
(6)
(7)

Based on literature [11], the number of kilomoles of exhaust gases per each kilogram of fuel can be expressed with relationships (8), (9), (10) and (11):

-carbon dioxide

$$n_{co_2} = \frac{w_c}{m_c} , \frac{kmol}{kg}$$
(8)

-nitrogen

$$n_{N_2} = 0.79 \cdot N_{air.act} , \frac{kmol}{kg}$$
(9)

-oxygen

$$n_{O_2} = 0.21 \cdot (\lambda - 1) \cdot N_{air} , \frac{kmol}{kg}$$
(10)

-water

$$m_{H_2O} = \frac{w_c}{w_h}, \frac{kmol}{kg}$$
(11)

Mole flow rate of exhaust gases (12) is the product of exhaust gas components and the flow rate of combusted fuel.

$$\dot{m}_i = n_i \cdot \dot{m}_{fuel} , \frac{kmol}{s}$$
(12)

However, mass flow rate of exhaust gases (13) is the product of the mole flow rate and the molar mass of exhaust gases.

$$\dot{M}_i = \dot{m}_i \cdot m_i \,, \frac{kg}{s} \tag{13}$$

Total flow rate of exhaust gases (14) is the sum of individual flow rates of exhaust gas components.

$$\dot{M}_{ex} = \sum \dot{M}_i \,, \frac{kg}{s} \tag{14}$$

In order to find mean specific heat of exhaust gases, it is necessary to calculate mass fractions of exhaust gas components (15) for each combustion product, which is the product of the component's mass flow rate and the mass of exhaust gases.

$$g_i = \frac{\dot{M}_i}{\dot{M}_{ex}} \tag{15}$$

The calculation results are presented in Table 4.

2.2. Thermodynamic calculations

The thermodynamic calculations were based on the ideal Otto cycle which describes the functioning of a spark ignition piston engine. The cycle consists of two adiabats (1-2, 3-4) and two isochores (2-3, 4-1). It describes the following principle of operation: adiabatic compression of the aspirated mixture occurs from point 1 to point 2,

while isochoric combustion of fuel occurs from point 2 to point 3. Increasing pressure causes the piston to move, thus effecting an adiabatic expansion of exhaust gases (points 3-4). They cycle ends (points 4-1) with an isochoric removal of heat. The cycle has additional charge exchange, as represented by points 1-0-1.

As the physical properties of the fuel/air mixture are slightly different than the properties of pure air, the calculations were simplified by assuming that the working medium is pure air. The assumed normal conditions were p_1 =1013525 Pa, T_1 =25°C and the atomic mass was μ_{air} =0.02889 g/kmol.

Specific volume in point 1 (16) was calculated from the Clapeyron equation.

$$v_1 = \frac{RT_1}{\mu_{air} \cdot p_1} , \frac{m^3}{kg}$$
(16)

Conditions in point 2 were calculated on the basis of the compression ratio ε of the BMW M50B20 engine, from (17).

$$v_2 = \frac{v_1}{\varepsilon}, \frac{m^3}{kg}$$
(17)

By assuming adiabatic exponent κ =1.4, pressure and temperature in point 2 were calculated from equations (18) and (19).

$$p_2 = p_1 \cdot \varepsilon^{\kappa} , Pa$$
(18)
$$T_2 = T_1 \cdot \varepsilon^{\kappa-1} , K$$
(19)

Prior to calculating pressure in point 3, it was necessary to find temperature T_{3} . This was done with equation (20) as described in [12].

$$T_3 = T_1 + \frac{W_d}{(1+\lambda \cdot L_{tp}) \cdot c_{p.ex}} , K$$
(20)

Calculating temperature from equation (20) required finding specific heat of exhaust gases. For this purpose, mean specific heat values at constant pressure were calculated (21), in accordance with [11].

$$c_{p.ex} = \sum (g_i \cdot c_{p_i}), \frac{kJ}{kg \cdot K}$$
(21)

Air consumption per one kilogram of combusted fuel was calculated from equation (22), described in [12].

$$L_{tp} = \frac{1}{0.232} \left(\frac{8}{3} \cdot w_c + 8 \cdot w_h \right) , \frac{kg \, air}{kg \, fuel}$$
(22)

Subsequently, pressure in point 3 was determined using isochoric equation (23).

$$p_3 = \frac{p_2 \cdot p_3}{T_2}$$
, *Pa* (23)

Pressure and temperature in point 4 were calculated from adiabatic equations (24) and (25). The total flux of energy lost in the exhaust gases is expressed in relationship (26).

$$p_4 = \frac{p_3 \cdot v_3^\kappa}{v_4^\kappa} , Pa$$
 (24)

Tab. 4. Calculated parameters for individual rotational speeds

							1			
Deremeter	Linit	Rotational speed, rev/min								
Falailletei	Unit	1500	1750	2000	2250	2500	2750	3000	3250	3500
<i>V</i> _{air}	m³/s	0.029	0.029	0.033	0.038	0.042	0.046	0.050	0.054	0.058
\dot{m}_{fuel}	kg/s · 10-3	2.404	2.404	2.747	3.091	3.434	3.777	4.121	4.464	4.808
n_{CO_2}	kmol/kg		0.071							
n_{N_2}	kmol/kg		0.428							
n_{O_2}	kmol/kg		0.005							
n_{H_2O}	kmol/kg		0.075							
М _{CO2}	kg/s	0.007	0.007	0.008	0.009	0.011	0.012	0.013	0.014	0.015
\dot{M}_{N_2}	kg/s	0.029	0.029	0.033	0.037	0.041	0.045	0.049	0.053	0.058
М ₀₂	kg/s	0.001	0.001	0.001	0.001	0.001	0.001	0.001	0.001	0.001
	kg/s	0.003	0.003	0.003	0.004	0.004	0.005	0.005	0.006	0.006
M _{ex}	kg/s	0.040	0.040	0.046	0.051	0.057	0.063	0.068	0.074	0.080
g_{CO_2}	-	0.188								
g_{N_2}	-		0.721							
g_{O_2}	-		0.010							
g_{H_2O}	-	0.081								
$\dot{Q}_{ m ex}$	kW	35.7	35.7	40.8	45.9	51.1	56.2	61.3	66.4	71.5

$$T_4 = \frac{T_3}{\left(\frac{\nu_4}{\nu_3}\right)^{\kappa-1}} , K$$
 (25)

$$\dot{Q}_{ex} = \dot{M}_{ex} \cdot c_{p.ex} \cdot (T_4 - T_1), W \tag{26}$$

Tab. 5. Parameters in the characteristic points of the Otto cycle

Daramator	Llnit	Point					
Falametei	Unit	1	2	3	4		
p	MPa	0.101	2.545	8.567	0.341		
V	m³/kg	0.847	0.085	0.085	0.847		
Т	K	298.2	748.9	2520.7	1004.5		

Table 5 shows the parameters of the working medium used in the combustion engine. The temperature calculated in point 4 is the temperature of exhaust gases escaping through the exhaust system. Its value of approximately 730°C is consistent with the values proposed in literature [10, 11, 12]. Table 6 shows specific heat values of exhaust gases and mean specific heat values of exhaust gas components for the 298-3500 K temperature range. The flux of heat lost with the exhaust gases for various rotational speeds is provided in Table 4.

 Tab. 6. Specific heat values of exhaust gases and mean specific heat values of exhaust gas components for the temperature range of 298-3500 K [11]

Parameter	Parameter Unit		Exhaust gas component					
i arameter	Unit	CO ₂	O ₂	H ₂ O	N ₂			
Cp	kJ/kg ⋅K	1.051	1.095	2.518	1.187			
C _{p.ex}	kJ/kg ⋅K	1.268						

Heat exchange between the exhaust gases and the thermoelectric generator

The calculations were based on the 2411G-7L31-15CX1 thermoelectric module manufactured by Custom thermoelectric and shown in Fig. 4.



Fig. 4. Overall dimensions of the thermoelectric module [13]

The module is made of a PbTe compound, and its parameters and overall dimensions are presented in Table 8.

 Tab. 8. Parameters of the Customthermoelectric 2411G-7L31-15CX1 module [13]

Parameter	Unit	Value
а	m	0.056
b	m	0.056
Ate	m ²	3.136·10 ⁻³
δ _{te}	m	0.005
λ_{te}	W/m·K	2.18
Tmaxte	K	593.15

In order to determine the amount of heat absorbed by the thermoelement, volumetric exhaust gas flow was calculated from equation (27), and exhaust gas density was determined by transforming the Clapeyron equation (28).

$$\dot{V}_{ex} = \frac{\dot{M}_{ex}}{2} , \frac{m^3}{2}$$
(27)

$$\rho_{ex} = \frac{\frac{p_{ex}}{p_4}}{\frac{k_g}{R_2 T_4}}, \frac{k_g}{m^3}$$
(28)

Equivalent gaseous constant was calculated from (29).

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$$R_Z = 8.3144 \cdot \sum \frac{g_i}{m_i} , \frac{J}{kg \cdot \kappa}$$
(29)

The overall dimensions required to perform initial calculations were determined by the thermoelectric module. It was assumed that the generator will consist of the hot layer, cold layer and thermoelectric layer. The hot layer will consist of channels accommodating exhaust gases; the cold layer – of channels accommodating the coolant; the thermoelectric layer will comprise thermoelectric cells. Based on literature [2, 14, 15], it was assumed that the cold side of the thermoelement can be kept at temperature T_c =50°C. The number of generator rows was due to the amount of electric energy collected by the modules and to the value of the counterpressure. Exhaust gases were assumed to flow through three channels, each having height u = 5 mm. Each row was decided to accommodate 6 thermoelements (Fig. 5)



Fig. 5. Schematic diagram of the thermoelectric generator

Exhaust gas velocity in each channel was calculated from (30).

$$w_{ex} = \frac{V_{ex}}{3 \cdot (a \cdot u)} , \frac{m}{s}$$
(30)

Equivalent hydraulic diameter was calculated on the basis of equation (31).

$$d_h = 4 \cdot \frac{A_k}{O_k}, m \tag{31}$$

Heat conductivity coefficient was based on the temperature of the medium and on literature [16] and calculated from relationships (32), (33), (34) and (35).

$$\lambda_{CO_2} = 1.41 \cdot 10^{-2} + 7.6444 \cdot 10^{-5} T_4 + 3.7778 \cdot 10^{-8} T_4^2 - 8.8889 \cdot 10^{-11} T_4^3, \frac{W}{m_{*}K}$$
(32)

$$\lambda_{H_20} = 1.7 \cdot 10^{-2} + 5.6983 \cdot 10^{-5} T_4^{+} + \tag{33}$$

$$1.297172 \cdot 10^{-7} T_4^2 - 9.13131 \cdot 10^{-11} T_4^3, \frac{W}{m \cdot K}$$

$$\lambda_2 = 2.38 \cdot 10^{-2} + 9.4414 \cdot 10^{-5} T_4 + 10^{-5} T_5 +$$

$$-1.352525 \cdot 10^{-7} T_4^2 - 1.8989 \cdot 10^{-10} T_4^3, \frac{W}{m \cdot K}$$

$$\lambda_{N} = 2.43 \cdot 10^{-2} + 7.4717 \cdot 10^{-5} T_4 + 3.5727 \cdot 10^{-5} T_5 + 3.5727 \cdot 10^{-5} \cdot 10^{-$$

$$-7.1616 \cdot 10^{-8} T_4^2 + 6.8686 \cdot 10^{-11} T_4^3, \frac{W}{m_1}$$
(35)

Mean heat conductivity coefficient was determined from equation (36).

$$\lambda_{ex} = \sum g_i \cdot \lambda_i \; , \frac{W}{m \cdot K} \tag{36}$$

Using data provided in [16] as well as equations (37), (38), (39) and (40), dynamic viscosity coefficients were determined for exhaust gas components, in a manner analogous to the heat conductivity coefficients.

$$\eta_{CO_2} = 1.37 \cdot 10^{-5} + 4.7033 \cdot 10^{-8} T_4 + -2.5666 \cdot 10^{-11} T_4^2 + 1.3333 \cdot 10^{-14} T_4^3, Pa \cdot s$$
(37)

$$\eta_{H_20} = 9.60 \cdot 10^{-6} + 2.7813 \cdot 10^{-8}T_4 +$$

$$4.6260 \cdot 10^{-11}T_2^2 = 5.0545 \cdot 10^{-14}T_3^3 P_{23} \cdot c$$
(38)

$$\eta_{02} = 1.88 \cdot 10^{-5} + 6.0260 \cdot 10^{-8} T_4^{+}$$

$$-6.1272 \cdot 10^{-11}T_4^2 + 5.5757 \cdot 10^{-14}T_4^3, Pa \cdot s$$
⁽³⁹⁾

$$\eta_{N_2} = 1.67 \cdot 10^{-5} + 4.3805 \cdot 10^{-8}T_4 +$$
(40)

$$2.0494 \cdot 10^{-11}T_4^2 + 6.8686 \cdot 10^{-15}T_4^3$$
, $Pa \cdot s$

Mean kinematic viscosity was calculated from (41). $\sum_{n=1}^{\infty} a_{n} m^{2}$

$$v_{ex} = \frac{2 g_i \eta_i}{\rho_{ex}} , \frac{m}{s}$$
(41)

Reynolds number for exhaust gases flowing through the channel was found on the basis of (42).

$$Re_{ex} = \frac{w_{ex} \cdot \lambda_{ex}}{d_h} \tag{42}$$

The flow was turbulent for each rotational speed, and therefore, as suggested in literature [17], the Nusselt number was calculated from equation (43), and exhaust gas heat transfer coefficient was calculated from (44).

$$Nu_{ex} = 0.021 Re_{ex}^{0.8} \cdot Pr_{ex}^{0.43}$$
(43)

$$h_{ex} = \frac{m_{ex} m_{ex}}{d_h}, \frac{m^2 K}{m^2 K}$$
(44)

In order to perform further calculations, it was assumed that the structural components of the generator are made of stainless steel St 44K (P355GH), which displays significant resistance to high temperatures. The assumed wall thickness was δ_p =0.002 m. The material has heat conductivity coefficient λ_p =50 W/(m·K). Fig. 6 shows the distribution of temperatures and heat fluxes.



Fig. 6. Distribution of temperatures and heat fluxes for a fragment of the exhaust gas channel surrounded with thermoelements

The calculations were primarily aimed at finding temperature T_H on the hot side of the thermoelement, so as to determine the amount of electric energy generated by the module. Prior calculations, however, should provide the total flux of heat transferred from the exhaust gases through the wall to the thermoelement. Heat exchange takes place through forced convection. T_{1f} is the temperature of exhaust gases, T_{1p} is the temperature of the wall, T_H is the unknown temperature on the hot side of the module, T_C is the temperature on the cold side of the module, as based on literature [2, 14, 15].

Total heat flux was calculated on the basis of the Fourier law and Newton's law, and is expressed with relationship (45).

$$\dot{Q}_{all} = \frac{T_{1f} - T_C}{\frac{1}{h_{ex}} + \frac{\delta p}{\lambda_p} + \frac{\delta t_e}{\lambda_{te}}} \cdot A_{te} , W$$
(45)

According to the heat transfer law, the fluxes in each of the walls are equal, and thus equation (46) is valid.

$$\dot{Q}_1 = \dot{Q}_2 = \dot{Q}_{all} \tag{46}$$

Based on this relationship, the temperature of the wall and of the exhaust gas channel was calculated from (47).

$$T_{1p} = T_{1f} - \frac{Q_1}{h_{ex}A_{te}} , K$$
 (47)

Observing the $T_{H} < T_{max.te}$ condition, temperature on the hot side of the thermoelement was calculated from equation (48).

$$T_H = T_{1p} - \frac{Q_2 \cdot \delta_p}{\lambda_p \cdot A_{te}} , K$$
(48)

Using the power characteristic of the thermoelement as a function of the temperature on the cold side and on the hot side of the thermoelement (Fig. 7), relationship (49) was established, which allows finding the generated electric power.



Fig. 7. Power characteristic of the thermoelement as a function of the cold and hot temperature [13]

$$P_{el.te} = 6.526 \cdot 10^{-14} T_{H}^{6} - 1.777 \cdot 10^{-10} T_{H}^{5} + 1.992 \cdot 10^{-7} T_{H}^{4} - 0.001 T_{H}^{3} + 0.038 T_{H}^{2} + -6.894 T_{H} + 509.137. W$$
(49)

Relationship (50) describes heat flux transferred from exhaust gases in each of the three channels of the nth row.

$$Q_{ex.n1.3} = Q_{ex.2.3} = Q_{ex.3.3} = = \frac{M_{ex}}{3} \cdot c_{p.ex} \cdot (T_{1f} - T_{amb}), W$$
(50)

Considering the fact that exhaust gas heat flux is absorbed by the upper and by the lower wall of the channel, heat flux in the subsequent row was calculated from (51).

$$\dot{Q}_{ex.(n+1)\ 1.3} = \dot{Q}_{ex.n1.3} - 2 \cdot \dot{Q}_1, W$$
 (51)

Thus, exhaust gas temperature in the n+1 row was calculated from relationship (52).

$$T_{1fn+1} = \frac{Q_{ex.n1.3} - 2Q_1}{\frac{\dot{M}_{ex}}{3} c_{p.ex}} , K$$
 (52)

The temperature decrease between the nth row and the n+1 row is expressed with equation (53).

$$\Delta T_n = T_{1fn} - T_{1fn+1} , K$$
 (53)

The total electric energy produced in one row is a sum from all modules in that row and is expressed with (54).

$$P_{el} = 6 \cdot P_{el.te} , W \tag{54}$$

Pressure losses

The number of rows in the generator was determined on the basis of the loss of pressure generated by the TEG. Literature studies indicate [2, 18] that the projects generated pressure losses of 30-65 millibars, and thus 65 mbar was assumed to be the upper limit determining the number of thermoelement rows. Pressure drop was calculated on the basis of [19]. For this purpose, equation (55) was used to determine equivalent diameter (Fig. 8) for the three channels.

$$d_{h3} = \sqrt{3} \cdot d_h \tag{55}$$

The assumed exhaust pipe diameter being d_{in} =0.047 m, relationship (56) served to calculate exhaust gas speed.

$$w_{ed} = \frac{V_{ex}}{\frac{\pi d_W^2}{4}}, \frac{m}{s}$$
(56)



Fig. 8. An illustration of the narrowing formed by the TEG

Based on the data provided in [19], both local loss coefficient and linear loss coefficient were determined. Local loss coefficient on the inlet was calculated from (57). The coefficient on the outlet was assumed to be ξ_2 =1, and linear loss coefficient was assumed to be λ_{sl} =0.03.

$$\xi_1 = 0.5 \cdot \left(1 - \left(\frac{d_{h3}}{d_{in}}\right)^2\right) = 0.443 \tag{57}$$

Both local and linear losses were calculated on the basis of [19], from relationships (58) and (59).

$$\Delta h_m = \xi \cdot \frac{w_{ed}^2}{2 \cdot g}, m \tag{58}$$

$$\Delta h_l = \lambda_{sl} \cdot \frac{5 \cdot a}{d_{h3}} \cdot \frac{w_{ex}^2}{2 \cdot g} , m$$
(59)

Total pressure loss (60) is the sum of individual losses.

$$\Delta h = \Sigma \Delta h_m + \Sigma \Delta h_l , m \qquad (60)$$
Pressure drop is described with equation (61).

$$\Delta p = \rho_{ex} \cdot g \cdot \Delta h , Pa \tag{61}$$

3. ANALYSIS OF THE RESULTS

After all of the calculations had been performed, it was decided that the generator would consist of 5 thermoelement rows, 6 modules each. This number was due to the 63 mbar counterpressure, which was observed at 3500 rev/min. Hence, the total number of 30 modules provide 450 W of electric energy at the rotational speed of 2000 rev/min. Based on an assumption that the engine is installed in a medium-size car whose electric energy consumption is approximately 750 W, the modules provide 60% of the required energy. Additional parameters are shown in Fig. 9.

The maximum allowable temperature on the hot side of the thermoelectric module is 593 K. The black line in Fig. 9 represents the limit, above which the T_{H1} temperatures on the hot side of the module (in the first row) exceed the $T_{max.te}$ temperature. The authors decided not to modify the calculations performed for higher rotational speeds for two reasons. The first reason is that producing much electric energy already in the lowest rotational speeds of the combustion engine is an advantageous solution. The second reason is to demonstrate the gains expected from a thermoelectric generator. Thermoelectric materials are being continuously developed and modules having higher efficiency and operating temperatures above 320°C (593 K) will become commercially available in the future.

One of the methods to prevent a thermoelectric generator from overheating is to employ a special bypass. In the case when the mass flow of exhaust gases is excessive, such a device would allow the gases to be diverted around the generator and sent directly to the atmosphere, simultaneously maintaining the temperature on the hot side of the module within limits. Solutions of this type are offered by a number of companies which are engaged in research on thermoelectric generators applied in vehicles.

The η_{TEG} efficiency, i.e. the ratio between heat transferred through the walls and electric energy produced by the modules, was approximately 24-26%. However, if the generated electric power is viewed in relation to the energy of the exhaust gases, the total efficiency did not exceed 1.1%. This limitation is dictated by the still insufficient amount of heat transferred from the exhaust gases to the walls.

Fig. 10 shows the values of volumetric exhaust gas flow and the pressure drop generated by the system. All of the values are consistent with the tolerance limits suggested in literature [2, 18]. It is worth mentioning that with a bypass applied, the drops would be significantly limited above 2000 rev/min.

Fig. 11 is a visual representation of the analyzed generator, while Fig. 12 shows its layers.



Fig. 10. Volumetric exhaust gas flow and the pressure drop



Fig. 9. Resultant characteristics of the thermoelectric generator



Fig. 11. Thermoelectric generator – assembled



Fig. 12. Thermoelectric generator – disassembled

SUMMARY AND CONCLUSIONS

This paper presents a calculation algorithm for analyzing the application of a thermoelectric generator to the process of recovering waste heat from a combustion engine and converting it to useful energy. Additionally, by performing the calculations on a particular example, it was possible to determine the process and geometric parameters of the module, as these parameters are not offered in the relevant literature. The study was based on the model of a BMW M50B20 spark ignition engine and started with stoichiometric calculations. The results allowed to find mass flow rate of exhaust gases, which is between 0.034 kg/s and 0.080 kg/s within the range of 1500÷3500 rev/min. The increase of the flow rate was accompanied by an increase of heat flux lost to the atmosphere - between 30 kW and 70 kW. In the case of an engine having nominal power equal to 110 kW, losses in the exhaust system ranged from 27% to 63%. In the next step, the temperature in the exhaust system was determined, and its value proved consistent with the values offered in literature, i.e. approximately 730°C. This information served to calculate heat exchange. The thermoelectric module was selected from among the best commercially available devices. Its maximum operating temperature was 320°C. Based on the characteristics disclosed in the product data sheet, a relationship was deduced which allowed calculating the generated electric energy as a function of temperature on the hot and on the cold side of the module. The cold side temperature of the module was here assumed at 50°C. This value was dictated by the properties of the analyzed materials, in which this temperature level was possible to maintain. The assumed generator had a layer structure. Data offered in literature served to establish allowable pressure drop which can be generated by a system of this type, and the length of the exhaust gas channels in the generator was decided to correspond to five lengths of the selected thermoelectric module. Within the 1500÷2000 rotational speed range, the generator was capable of delivering between 325 W and 450 W of power. These values correspond respectively to 43% and 60% of the total electric energy demand in a car. The efficiency of the generator, in relation to the total energy lost in exhaust gases, was approximately 1%. Above 2000 rev/min, temperatures on the hot side of the thermoelement exceeded the maximum allowable value. In order to prevent the generator from overheating, further research may focus on designing a bypass for exhaust gases to be diverted around the generator. Future works may also be aimed at designing a cooling system for the complete generator, which will additionally enable more precise measurement of the generated energy.

NOMENCLATURE

- A_{te} surface of the thermoelectric module, m²
- c_p specific heat, kJ/kg K
- d_h equivalent diameter, m
- d_{in} inner diameter, m
- e electric charge, C
- E kinetic energy, J
- $E_{FA,FB}$ Fermi energy, eV
- g mass fractions of exhaust gas components
- i ratio of the number of cylinders in an in-line combustion engine
- L_{tp} air consumption, kg/kg
- m flywheel mass, kg

 \dot{m} – mole flow rate of exhaust gas components, kmol/s

- \dot{m}_{fuel} mass flow rate of fuel, kg/s
- Mo torque, Nm
- M mass flow rate of exhaust gas components, kg/s
- \dot{M}_{ex} total mass flow rate of exhaust gases, kg/s
- n_{co2} amount of CO_2 produced from fuel, kmol/kg
- n_{O2} amount of O₂ produced from fuel, kmol/kg
- n_{H2O} amount of H₂O produced from fuel, kmol/kg
- n_{N2} amount of N₂ produced from fuel, kmol/kg
- n rotational speed of combustion engine, rev/min
- N oxidizer to fuel ratio, kmol/kg
- Nu Nusselt number
- p pressure, Pa

 P_{el} – total electric energy produced by 1 row of thermoelements, W

- Pel.te electric power of the thermoelectric module, W
- $P_{\rm s}$ power of the combustion engine, kW
- r flywheel diameter, m
- R gaseous constant, J/kg K
- Re Reynolds number
- $\dot{Q}_{\rm all}$ total heat transfer through a surface, W
- $\dot{Q_1}$ heat flux from the exhaust gas to the wall, W
- \dot{Q}_2 heat flux from the channel wall to the thermoel. module, W
- $\dot{Q}_{ex.n1.3}$ exhaust gas heat flux in nth row, for 1st channel, W
- T_c temperature on the cold side of the module, K
- T_H temperature on the hot side of the module, K
- T_{1f} exhaust gas temperature, K
- T_{1p} temperature on the inside of the exhaust gas separator, K
- Tamb ambient temperature, K
- U voltage, V
- wex exhaust gas velocity in the exhaust gas channel, m/s
- w_{ed} exhaust gas velocity in the exhaust pipe, m/s
- W_d fuel calorific value, kJ/kg
- \dot{V}_{air} volumetric air flow, m³/s
- Vair actual air volume, m3/kg
- $V_{\rm sk}$ swept volume, dm³
- \dot{V}_{ex} volumetric exhaust gas flow, m³/s
- ZT non-dimensional quality factor of the thermoelectric module
- α Seebeck coefficient, μ V/K
- β thermoelectric power coefficient, -
- δ_{te} thickness of the thermoelectric module, m
- δ_p thickness of the channel wall, m

 h_{ex} – exhaust gas heat transfer coefficient, W/m² K

 Δh_m – local losses, m

- Δh_l linear losses, m
- Δp pressure drop, Pa
- ΔT_n temperature drop between the rows, K
- $\varepsilon~$ compression ratio
- ε_T thermoelectric power, V
- $\epsilon_{2A1,1B2}$ electromotive power, V
- η dynamic viscosity, Pa s
- η_{all} total efficiency, %
- η_{te} efficiency of the thermoelectric module, %
- η_{TEG} efficiency of the thermoelectric generator, % κ adiabatic exponent
- λ_p heat conductivity coefficient of the gas separator, W/m K
- λ_{sl} linear loss coefficient, -
- λ_{te}^{*} heat conductivity coefficient of the thermoelement, W/m K v specific volume, m³/kg
- v_{ex} kinematic viscosity of the exhaust gas, m²/s
- ξ local loss coefficient, -

 ρ_{ex} – exhaust gas density, kg/m³

- σ electric conductivity, S
- au number of the cylinder
- ω rotational speed of the flywheel, rad/s

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Analiza systemu wykorzystującego generator termoelektryczny do odzysku ciepła odpadowego z samochodu

W pracy przedstawiono algorytm obliczeniowy generatora termoelektrycznego do zastosowania w układzie wydechowym silnika samochodowego. W celu sprawdzenia poprawności metody obliczeniowej pokazano jej przydatność na wybranym przykładzie silnika spalinowego. Do obliczeń wykorzystano silnik marki BMW, a konstrukcję generatora oparto o prototyp tej samej marki. Przeprowadzono obliczenia obiegu cieplnego i wyznaczono parametry spalin na wylocie z silnika. Następnie wykonano obliczenia przekazywania ciepła od spalin do modułu termoelektrycznego i wyznaczono energię elektryczną uzyskiwaną z szeregu modułów. W ostatniej części przedstawiono wpływ prędkości obrotowej silnika na parametry pracy generatora termoelektrycznego.

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